

#### E190Q – Lecture 3 Autonomous Robot Navigation

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Figures courtesy of Siegwart & Nourbakhsh



#### **Control Structures Planning Based Control**





## **Motion Uncertainty**

- 1. Odometry & Dead Reckoning
- 2. Modeling motion
- 3. Odometry on the Jaguar
- 4. Example System



- Odometry
  - Use wheel sensors to update position
- Dead Reckoning
  - Use wheel sensors and heading sensor to update position
- Straight forward to implement
- Errors are integrated, unbounded



http://www.guiott.com



• Odometry Error Sources?



- Odometry Error Sources?
  - Limited resolution during integration
  - Unequal wheel diameter
  - Variation in the contact point of the wheel
  - Unequal floor contact and variable friction can lead to slipping



- Odometry Errors
  - Deterministic errors can be eliminated through proper calibration
  - Non-deterministic errors have to be described by error models and will always lead to uncertain position estimate.



## **Motion Uncertainty**

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• If a robot starts from a position p, and the right and left wheels move respective distances  $\Delta s_r$  and  $\Delta s_l$ , what is the resulting new position p'?





- To start, let's model the change in angle  $\Delta \theta$  and distance travelled  $\Delta s$  by the robot.
  - Assume the robot is travelling on a circular arc of constant radius.





Begin by noting the following holds for circular arcs:

$$\Delta s_l = R\alpha$$
  $\Delta s_r = (R+2L)\alpha$   $\Delta s = (R+L)\alpha$ 





Now manipulate first two equations:

$$\Delta s_l = R\alpha$$
  $\Delta s_r = (R+2L)\alpha$   
To:

$$R\alpha = \Delta s_l$$

$$L\alpha = (\Delta s_r - R\alpha)/2$$

$$= \Delta s_r/2 - \Delta s_l/2$$



• Substitute this into last equation for  $\Delta s$ :

$$\Delta s = (R+L)\alpha$$
  
=  $R \alpha + L\alpha$   
=  $\Delta s_l + \Delta s_r/2 - \Delta s_l/2$   
=  $\Delta s_l/2 + \Delta s_r/2$   
=  $\Delta s_l + \Delta s_r/2$   
=  $\Delta s_l + \Delta s_r/2$ 



 Or, note the distance the center travelled is simply the average distance of each wheel:





• To calculate the change in angle  $\Delta \theta$ , observe that it equals the rotation about the circular arc's center point





• So we solve for  $\alpha$  by equating  $\alpha$  from the first two equations:

$$\Delta s_l = R\alpha \qquad \Delta s_r = (R+2L)\alpha$$

This results in:

$$\Delta s_l / R = \Delta s_r / (R + 2L)$$

$$(R + 2L) \Delta s_l = R \Delta s_r$$

$$2L \Delta s_l = R (\Delta s_r - \Delta s_l)$$

$$\frac{2L \Delta s_l}{(\Delta s_r - \Delta s_l)} = R$$



Substitute R into

$$\alpha = \Delta s_l / R$$
  
=  $\Delta s_l (\Delta s_r - \Delta s_l) / (2L \Delta s_l)$   
=  $(\Delta s_r - \Delta s_l)$   
 $2L$ 

So...

$$\Delta \theta = \frac{(\Delta s_r - \Delta s_l)}{2L}$$



- Now that we have  $\Delta \theta$  and  $\Delta s_{j}$  we can calculate the position change in global coordinates.
  - We use a new segment of length  $\Delta d$ .





• Now calculate the change in position as a function of  $\Delta d$ .





Using Trig:

 $\Delta x = \Delta d \cos(\theta + \Delta \theta/2)$  $\Delta y = \Delta d \sin(\theta + \Delta \theta/2)$ 





• Now if we assume that the motion is small, then we can assume that  $\Delta d \approx \Delta s$ :





• Summary:

$$\Delta x = \Delta s \cos(\theta + \Delta \theta / 2)$$

$$\Delta y = \Delta s \sin(\theta + \Delta \theta / 2)$$
  

$$\Delta \theta = \frac{\Delta s_r - \Delta s_l}{b}$$
  

$$\Delta s = \frac{\Delta s_r + \Delta s_l}{2}$$
  

$$p' = f(x, y, \theta, \Delta s_r, \Delta s_l) = \begin{bmatrix} x \\ y \\ \theta \end{bmatrix} + \begin{bmatrix} \frac{\Delta s_r + \Delta s_l}{2} \cos(\theta + \frac{\Delta s_r - \Delta s_l}{2b}) \\ \frac{\Delta s_r + \Delta s_l}{2} \sin(\theta + \frac{\Delta s_r - \Delta s_l}{2b}) \\ \frac{\Delta s_r - \Delta s_l}{b} \end{bmatrix}$$



- Let's look at delta terms as errors in wheel motion, and see how they propagate into positioning errors.
  - Example: the robot is trying to move forward 1 m on the x axis.



$$\Delta s = 1 + e_s$$
$$\Delta \theta = 0 + e_\theta$$

where  $e_s$  and  $e_{\theta}$  are error terms



• According to the following equations, the error  $e_s = 0.001$ m produces errors in the direction of motion.

$$\Delta x = \Delta s \cos(\theta + \Delta \theta/2)$$
$$\Delta y = \Delta s \sin(\theta + \Delta \theta/2)$$

• However, the  $\Delta\theta$  term affects each direction differently. If  $e_{\theta} = 2 \text{ deg and } e_s = 0 \text{ meters}$ , then:  $cos(\theta + \Delta\theta/2) = 0.9998$  $sin(\theta + \Delta\theta/2) = 0.0175$ 



So

$$\Delta x = 0.9998$$
$$\Delta y = 0.0175$$

 But the robot is supposed to go to x=1,y=0, so the errors in each direction are

$$\Delta x = +0.0002$$

$$\Delta y = -0.0175$$

 THE ERROR IS BIGGER IN THE "Y" DIRECTION!



 Errors perpendicular to the direction grow much larger.





 Error ellipse does not remain perpendicular to direction.





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#### **Odometry on the Jaguar**

- Goals:
  - Calculate the resulting robot position and orientation from wheel encoder measurements.
  - Display them on the GUI.



#### **Odometry on the Jaguar**

- Method cont':
  - Make use of the fact that your encoder has resolution of 190 counts per revolution. Be able to convert this to a distance travelled by the wheel.

$$r\varphi_r = \Delta s_r$$

 Given the distance travelled by each wheel, we can calculate the change in the robot's distance and orientation.

$$\Delta s = \frac{\Delta s_r + \Delta s_l}{2} \qquad \qquad \Delta \theta = (\Delta s_r - \Delta s_l) \\ \frac{\Delta \theta}{2L}$$



#### **Odometry on the Jaguar**

- Method cont':
  - Now you should be able to update the position/ orientation in global coordinates.

$$\Delta x = \Delta s \cos(\theta + \Delta \theta/2)$$
$$\Delta y = \Delta s \sin(\theta + \Delta \theta/2)$$



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# The VideoRay MicroROV

#### **ROV Specs**

- Two horizontal thrusters, one vertical
- Forward facing color camera
- Rear facing B/W camera
- 1.4 m/s (2.6 knots) speed
- 152m depth rating
- Depth & Heading sensors
- SeaSprite Scanning Sonar





#### The VideoRay MicroROV

#### ROV Modeling

$$\begin{split} m[\dot{u} - vr + wq - x_G(q^2 + r^2) + y_G(pq - \dot{r}) + z_G(pr + \dot{q})] &= X \\ m[\dot{v} - wp + ur - y_G(r^2 + p^2) + z_G(qr - \dot{p}) + x_G(qp + \dot{r})] &= Y \\ m[\dot{w} - uq + vp - z_G(p^2 + q^2) + x_G(rp - \dot{q}) + y_G(rq + \dot{p})] &= Z \\ I_x \dot{p} + (I_z - I_y)qr - (\dot{r} + pq)I_{xz} + (r^2 - q^2)I_{yz} + (pr - \dot{q})I_{xy} \\ &+ m[y_G(\dot{w} - uq + vp) - z_G(\dot{v} - wp + ur)] &= K \\ I_y \dot{q} + (I_x - I_z)rp - (\dot{p} + qr)I_{xy} + (p^2 - r^2)I_{zx} + (qp - \dot{r})I_{yz} \\ &+ m[z_G(\dot{u} - vr + wq) - x_G(\dot{w} - uq + vp)] &= M \\ I_z \dot{r} + (I_y - I_x)pq - (\dot{q} + rp)I_{yz} + (q^2 - p^2)I_{xy} + (rq - \dot{p})I_{zx} \\ &+ m[x_G(\dot{v} - wp + ur) - y_G(\dot{u} - vr + wq)] &= N \end{split}$$



#### **Equations of Motion**

- 6 degrees of freedom (DOF):
- State vectors: body-fixed velocity vector: earth-fixed pos. vector:

$$\boldsymbol{\nu} = \begin{bmatrix} \boldsymbol{\nu}_1^T, \boldsymbol{\nu}_2^T \end{bmatrix}^T = \begin{bmatrix} u, v, w, p, q, r \end{bmatrix}^T$$
$$\boldsymbol{\eta} = \begin{bmatrix} \boldsymbol{\eta}_1^T, \boldsymbol{\eta}_1^T \end{bmatrix}^T = \begin{bmatrix} x, y, z, \phi, \theta, \psi \end{bmatrix}^T$$



DOF	Surge	Sway	Heave	Roll	Pitch	Yaw
Velocities	и	v	W	р	q	r
Position & Attitude	x	У	Z	$\phi$	θ	$\psi$
Forces & Moments	X	Y	Ζ	K	М	N



## **Equations of Motion**

#### **Initial Assumptions**

- The ROV will usually move with low velocity when on mission
- Almost three planes of symmetry;
- Vehicle is assumed to be performing non-coupled motions.

[W. Wang et al., 2006]



#### **Equations of Motion**

Horizontal Plane:

$$\begin{split} m_{11}\dot{u} &= -m_{22}vr + X_{u}u + X_{u|u|}u|u| + X\\ m_{22}\dot{v} &= m_{11}ur + Y_{v}v + Y_{v|v|}v|v|,\\ I\dot{r} &= N_{r}r + N_{r|r|}r|r| + N, \end{split}$$

Vertical Plan:

$$m_{33}\dot{w} = Z_w w + Z_{w|w|} w|w| + Z$$

[W. Wang et al., 2006]



## **Theory vs. Experiment**

- Coefficients for the dynamic model are pre-calculated using strip theory;
- A series of tests are carried out to validate the hydrodynamic coefficients, including
  - Propeller mapping
  - Added mass coefficients
  - Damping coefficients









#### **Propeller Thrust Mapping**

The forward thrust can be represented as:





#### **Direct Drag Forces**

 The drag can be modeled as non linear functions





#### **Perpendicular Drag Forces**

Heave (Z) drag from surge speed





#### **Model Verification**

#### Yaw Verification





#### **Model Verification**

#### Surge Verification





#### **Autonomous Control**

